

The 9th CCMC Workshop

# Plans for Implementation and Evaluation of KSEM L2 products

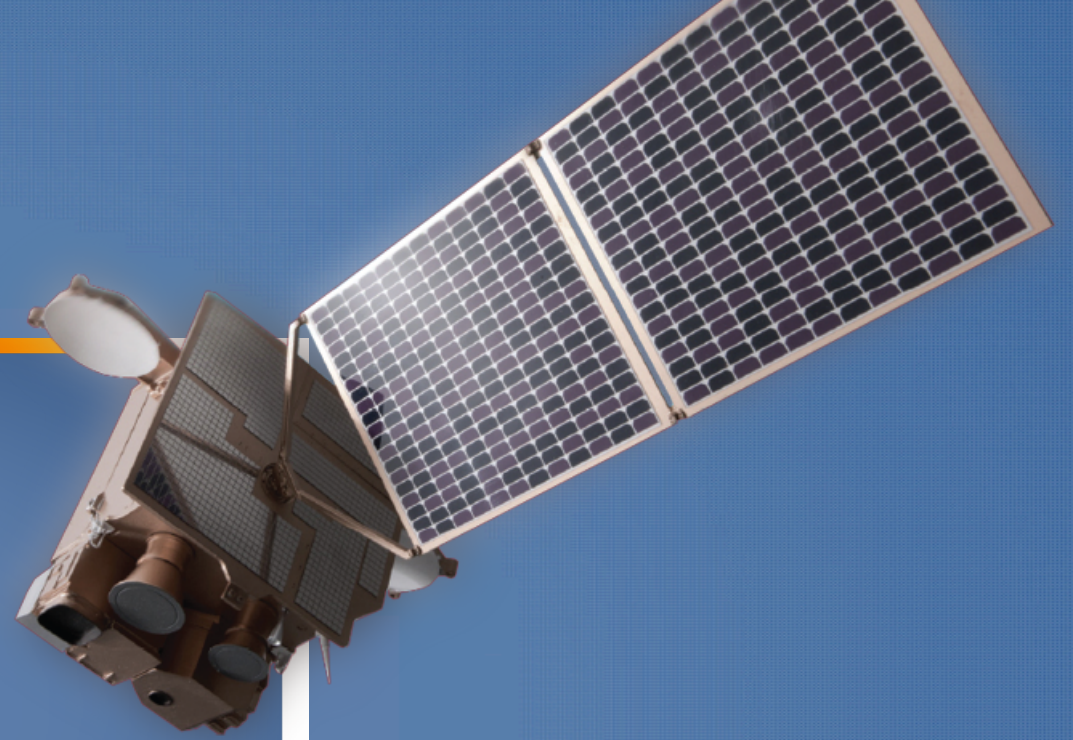
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Korea Meteorological Administration





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- I KSEM Project Overview
- II KSEM L2 Summary
- III Evaluation plan





# I GK2A, the Next Gen. Geo-Meteo Satellite

**GEO-KOMPSAT-2A** : Meteorological Satellite (To be launched in 2018)

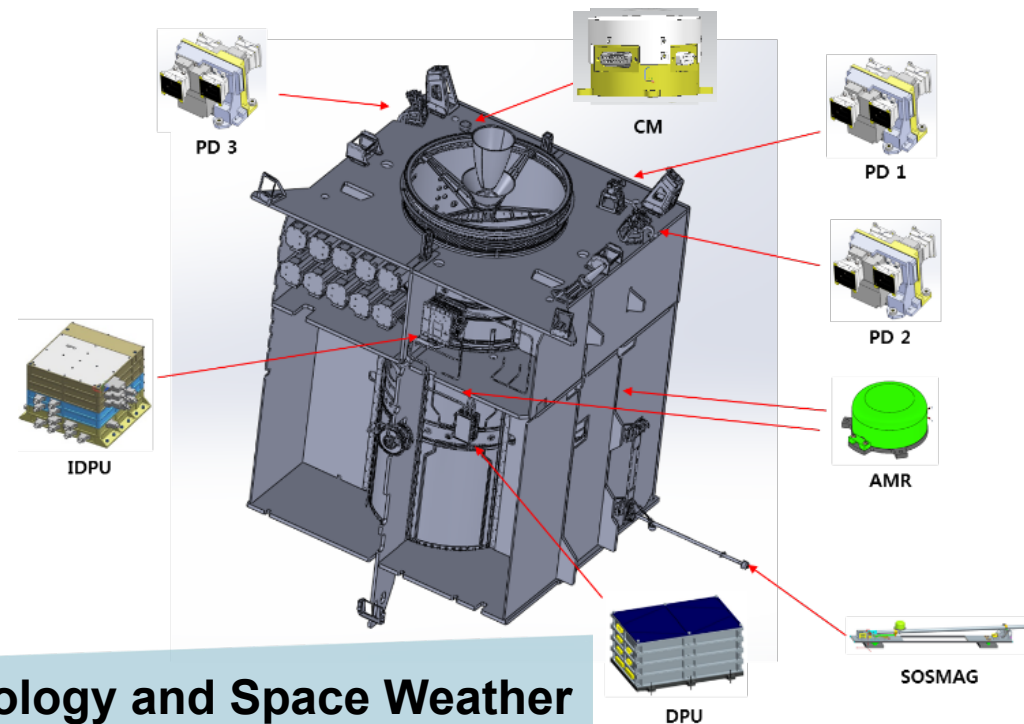
**GEO-KOMPSAT-2B** : Ocean/Environment compound satellite  
(To be launched in 2019)

## Advanced Meteorological Imager (AMI)



Korean Space wEather Monitor(KSEM)

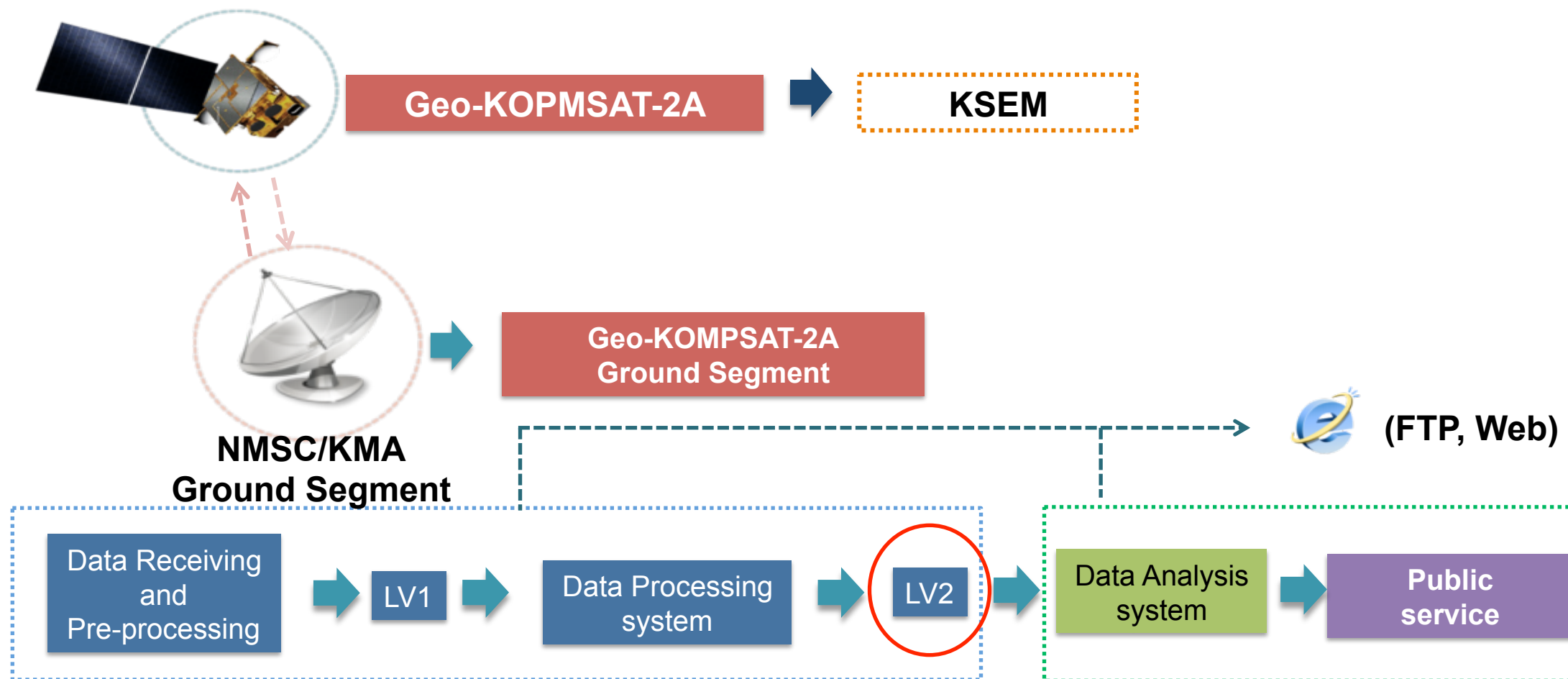
## KSEM accomodation on GK2A



## Meteorology and Space Weather

- ✓ Development Period : 2012 - 2018 (7 years)
- ✓ Orbit : 128.2°E over equator (36,000 km)
- ✓ Design life : 10 years

Sensor	Parameter	Requirement	Specification
PD	Energy Range	$100 \text{ keV} \leq E \leq 2 \text{ MeV}$	[Electron] $80 \text{ keV} \leq E \leq 2 \sim 3 \text{ MeV}$ [Proton] $80 \text{ keV} \leq E \leq \sim 10 \text{ MeV}$
	Energy Resolution	$\Delta E/E \leq 30\%$	[Electron] $25 \text{ keV} @ \sim 500 \text{ keV} / 50 \text{ keV} @ 1 \text{ MeV}$
			[Proton] $200 \text{ keV} @ 1 \text{ MeV} / 500 \text{ keV} @ 10 \text{ MeV}$
	Time Resolution	$\leq 0.33 \text{ sec}$	0.33 sec
	View Direction	5-direction	6-direction
	Geometric Factor	$\geq 10^{-3} (\text{cm}^2 \cdot \text{sr})$	0.02 ( $\text{cm}^2 \cdot \text{sr}$ )
	Background Contamination	$\leq 3\%$	$\leq 3\%$ for electrons $\leq 5\%$ for protons
	Count Resolution	$\geq 8 \text{ bit}$	8 bit
MG	Range	-350 nT ~ +350 nT (3-Axis)	Variable up to +/- 64,000 nT
	Accuracy	$\leq 1 \text{ nT}$	1 nT (after ground process)
	Time Resolution	$\leq 0.1 \text{ sec}$	0.1 s
	Type	Non-deployable	Deployable
CM	Range	-3 pA/cm <sup>2</sup> to +3 pA/cm <sup>2</sup>	-3 pA/cm <sup>2</sup> to +3 pA/cm <sup>2</sup>
	Accuracy	$\leq 0.01 \text{ pA/cm}^2$	0.0015 pA/cm <sup>2</sup>
	Time Resolution	$\leq 1 \text{ sec}$	1 sec



**LV1 : Reconstructed, Processed instrument data at full resolution, time-referenced annotated with ancillary information including calibration coefficients and geo-referencing parameters applied**

**LV2 : Product retrieved using additional algorithm or model with LV1 data**

**Level 1** : dissemination to the end user within **5 min.** after measurement

## Observation(Level 1) requiring space monitoring

**PD** :

High energy particle flux

**SOSMAG**:

Magnetic field in 3 axes(x, y, z)

**CM** :

Satellite internal charging



**Level 2** : dissemination to the end user within **30 min.** after **Level 1**

## Products (Level 2) requiring space monitoring (with 24hrs leading time)

**MPE**  
(Magnetospheric  
Particle Environment)

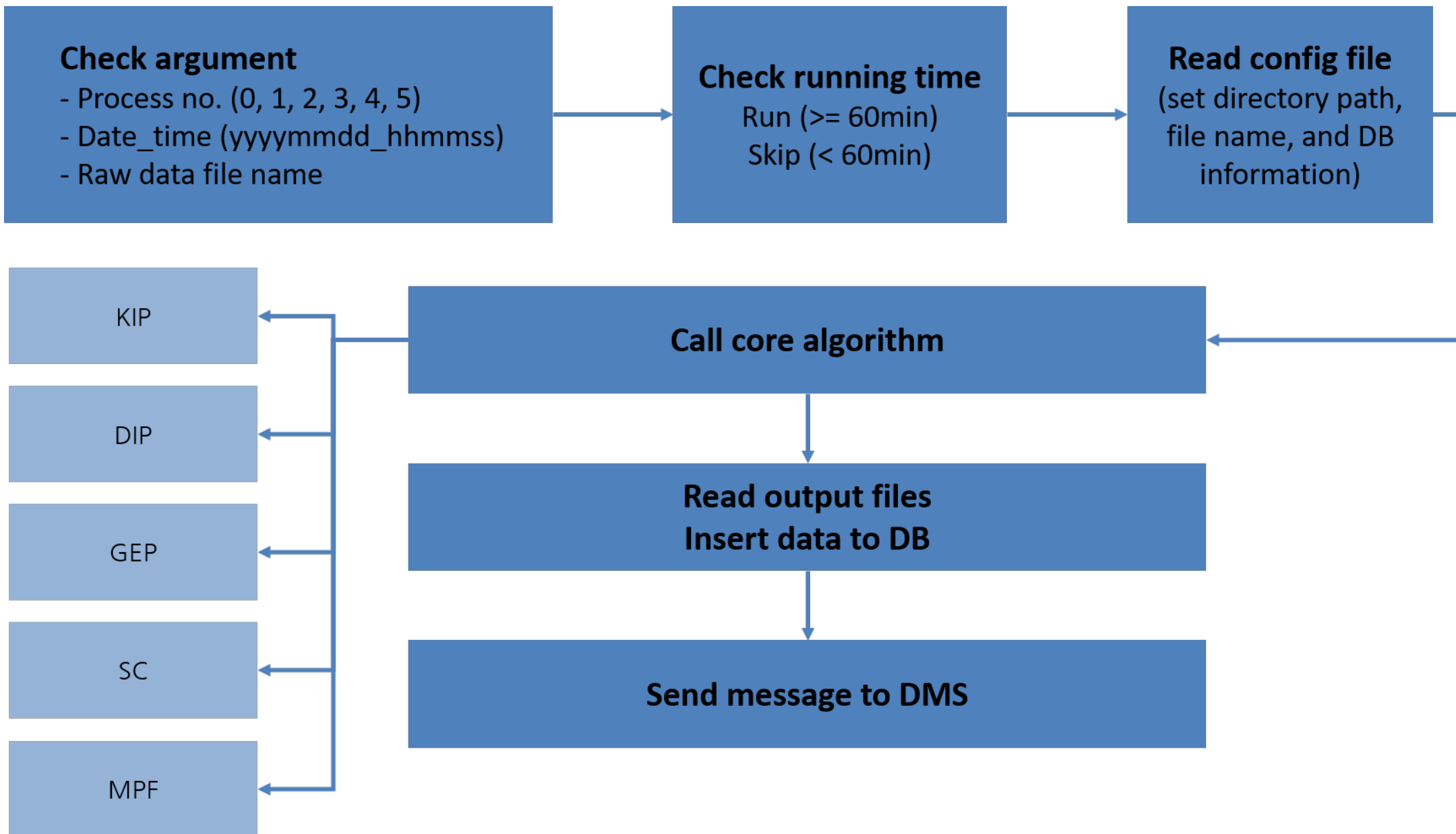
**GEP**  
(GK2A Electron flux Prediction)

**SC**  
(Satellite Charging monitor)

**DIP**  
(Dst Index Prediction)

**KIP**  
(Kp Index Prediction)

## II KSEM L2 Algorithm Architecture



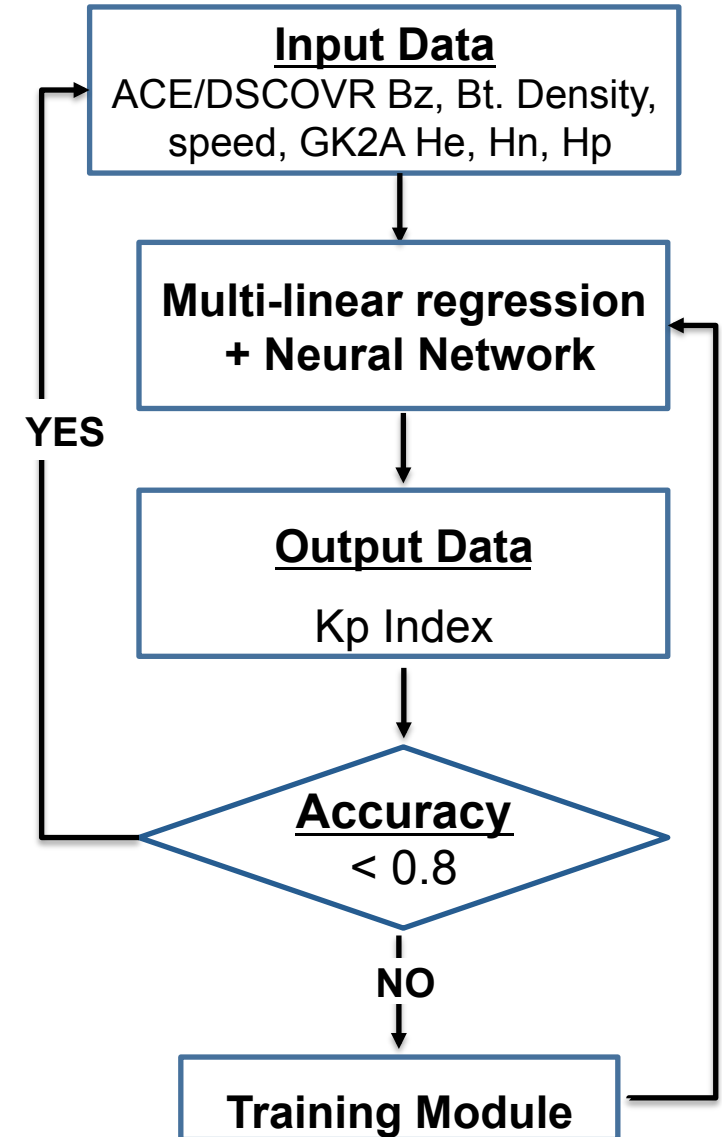
- **Product Description**

- Kp index prediction with 24 hrs leading time

- **Algorithm Description**

- Multi-linear regression + Neural network
  - **Multi-linear regression** made for solar wind data (speed, density and interplanetary magnetic field with 3 component of geomagnetic field)
  - **Artificial Neural Network (ANN)** for predicted Kp index using prediction results using multi-linear regression and solar wind data as the input data

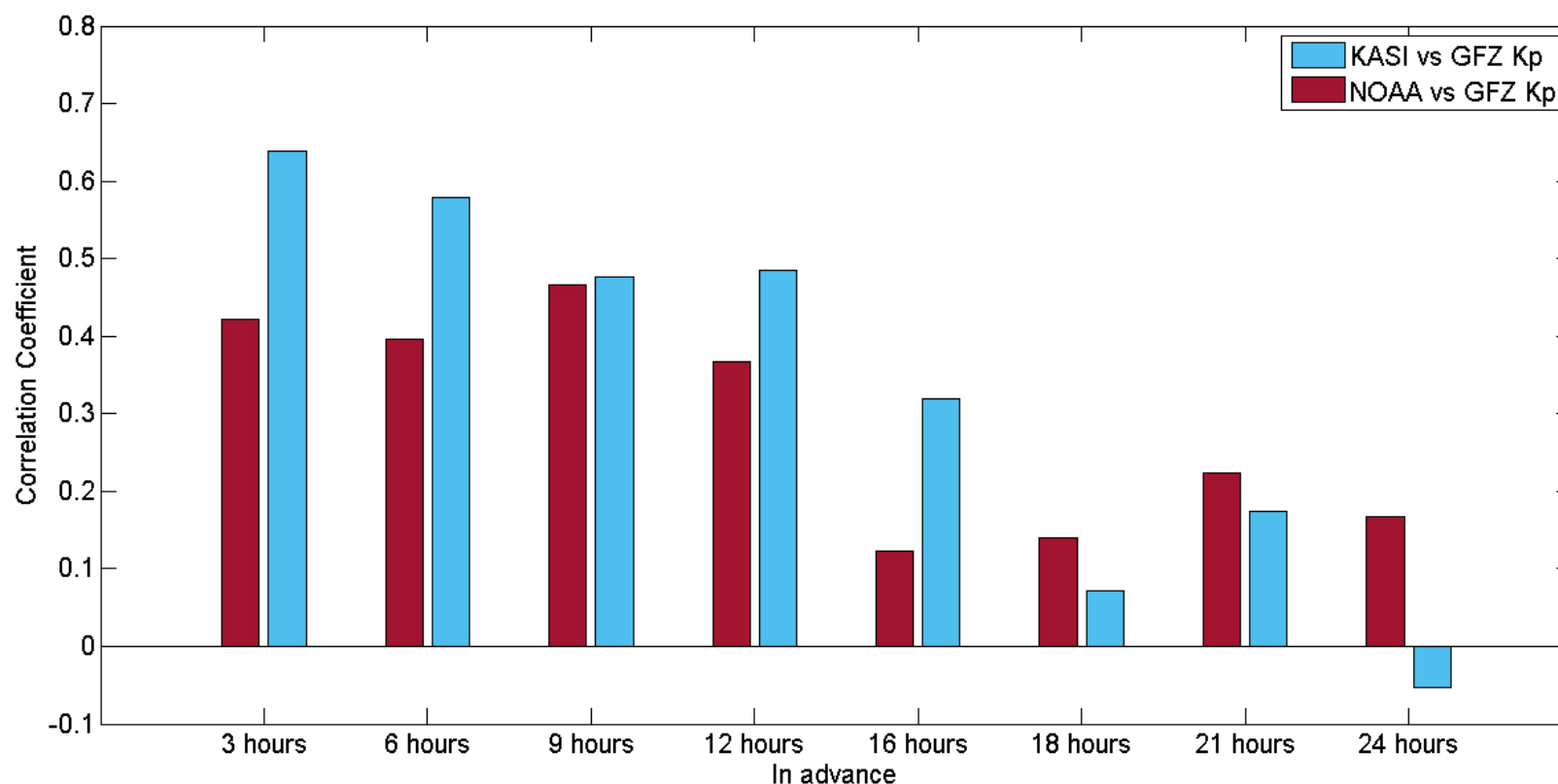
## [Flow Chart of Hybrid Algorithm]





## • Test Result

- For Algorithm validation, correlation coefficients between prediction and measurements are calculated for three months, from May 12 to July 15, 2017.
- The results are compared with NOAA predictions.
- The new model shows better prediction accuracy.

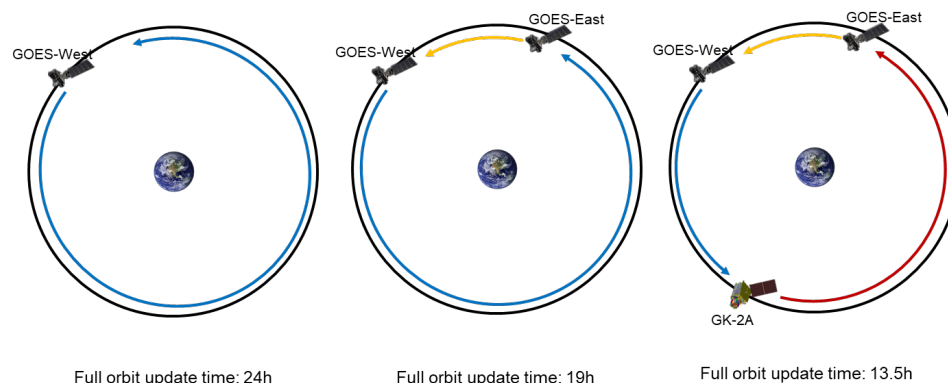


- **Product Description**

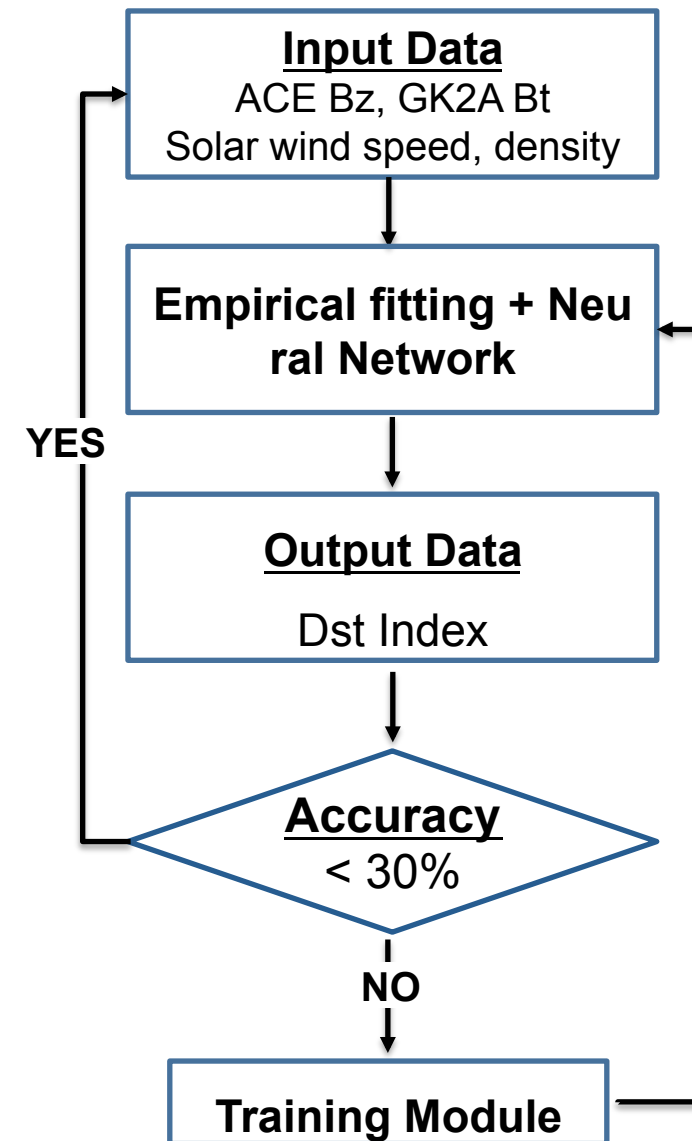
- Dst index prediction with 24 hours leading time

- **Algorithm Description**

- Empirical fitting + Neural network
  - **Empirical fitting** makes rough prediction from solar wind conditions
  - **Artificial Neural Network (ANN)** for predicting Dst index by using the empirical fitting, solar wind data and GEO magnetic field data.
    - ✓ This algorithm adopts multi-point magnetic field data to improve prediction accuracy.

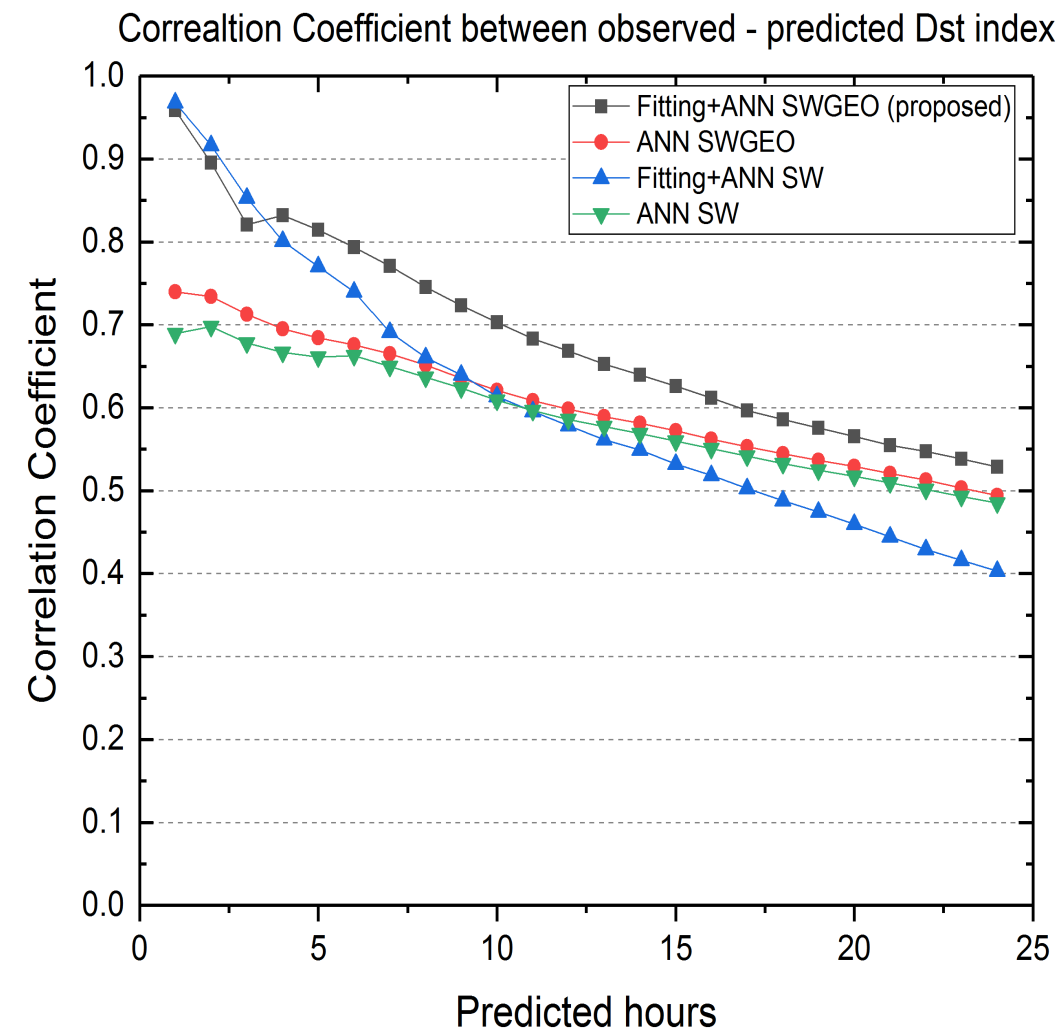


## [Flow Chart of Algorithm]



## • Test Result

- Test data from 2008 to 2015 used for performance validation of algorithms
- 4 algorithms were tested ;
  - **Fitting+ANN SWGEO** (Empirical fitting + artificial neural network with solar wind and geosynchronous data) Black
  - **ANN SWGEO** (Artificial neural network with solar wind and geosynchronous data) Red
  - **Fitting+ANN SW** (Empirical fitting+artificial neural network with solar wind data) Blue
  - **ANN SW** (Artificial neural network with solar wind data) Green

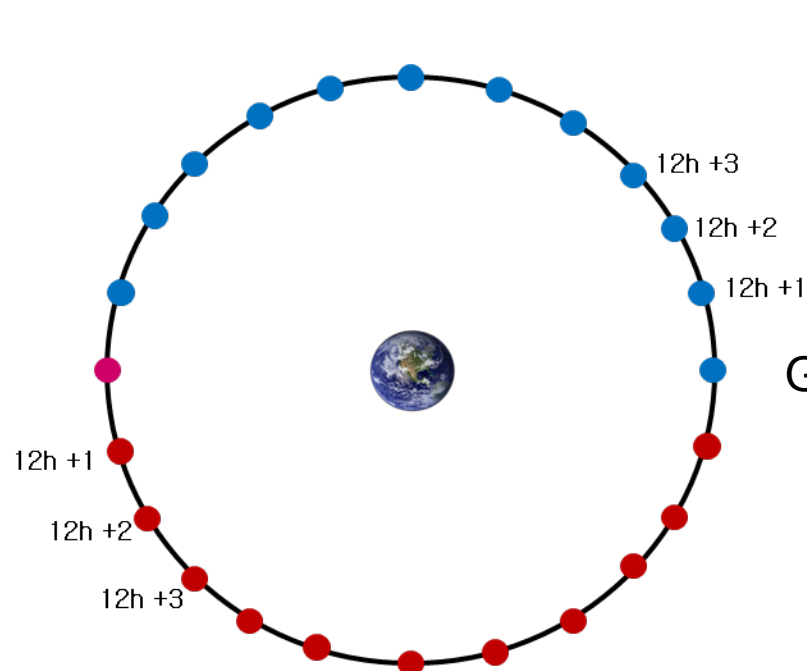


- **Product Description**

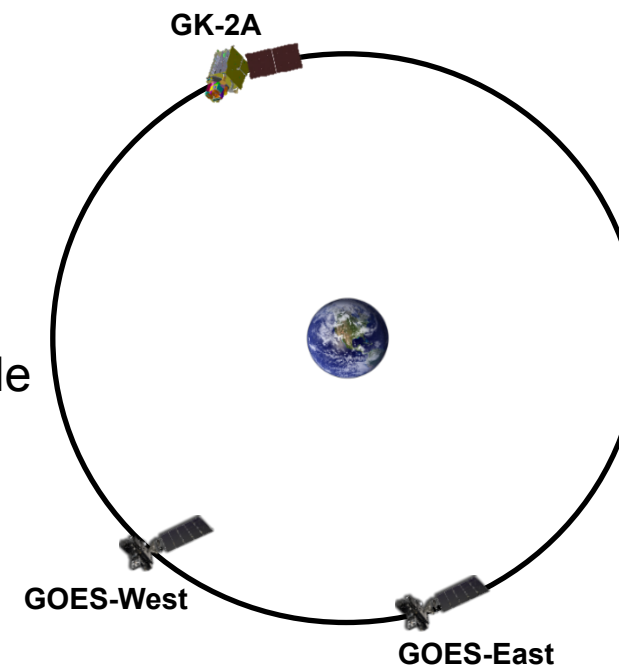
- Prediction of Electron flux for targeted geo-satellite with 24 hours leading time.  
(GK2A, FY-4 series, Himawari-series, GOES-series, MTG series)

- **Algorithm Description**

- **Neural network + multiple regression** with solar wind data from DSCOVR and geomagnetic index as input parameters and also using the data in combination with **GK-2A KSEM** and **NOAA GOES** electron flux.



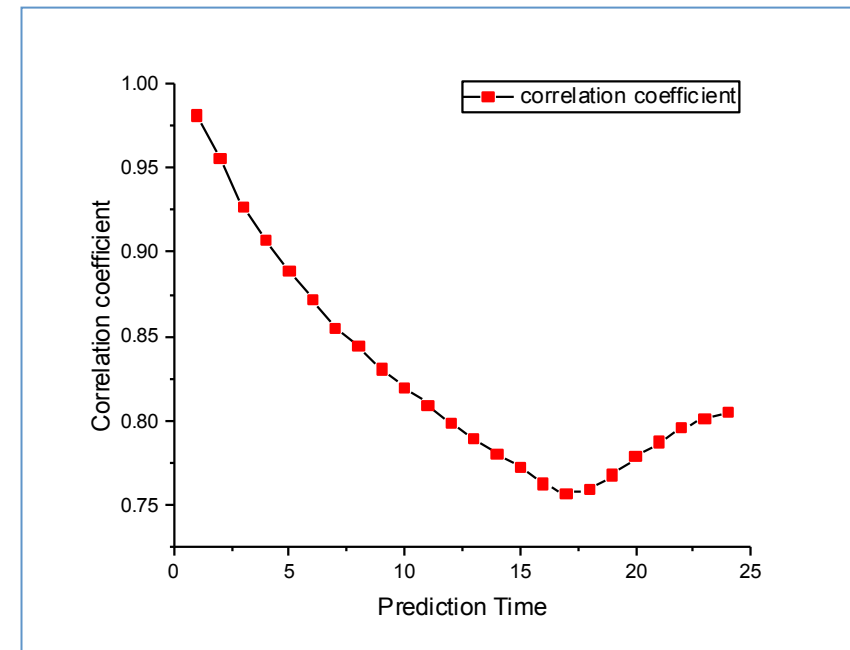
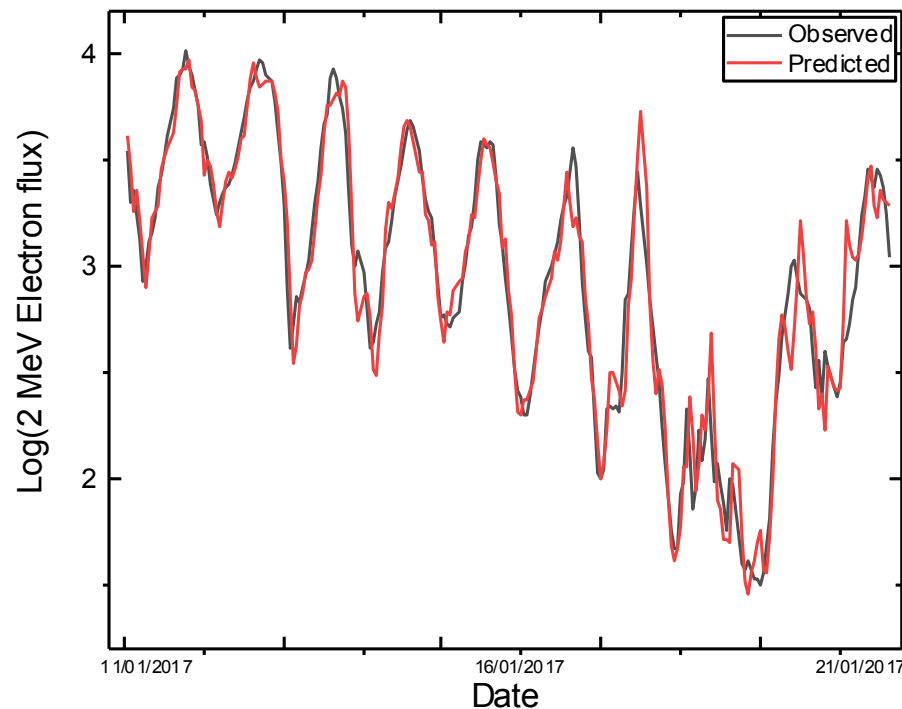
GK-2A locates almost opposite side of GOES-east





## • Test Result

- While old neural network adopts time-sequential input data from single satellite, new algorithm uses multipoint observation data as an input data.
- (Left panel) The prediction result with 1 hour leading time.
- (Right panel) correlation coefficient (y-axis) with the leading time(x-axis)



- **Product Description**

- Satellite charging index (internal current) with 24 hours leading time

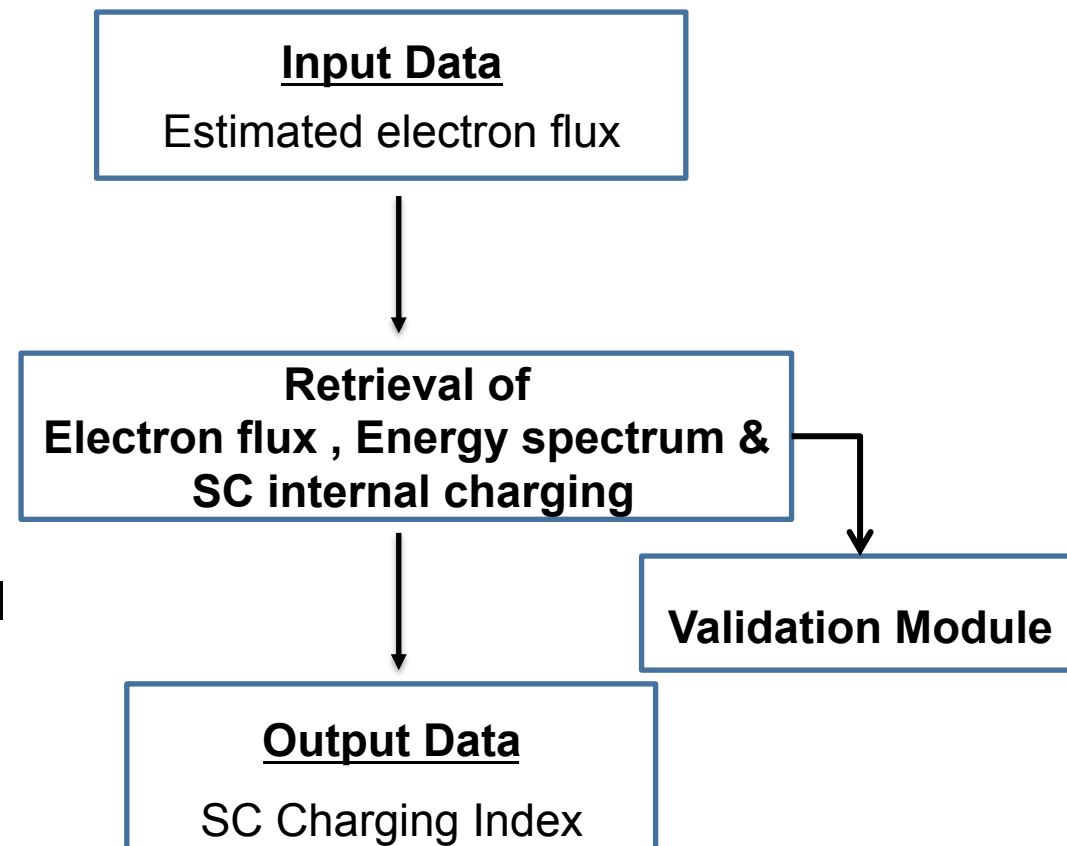
- **Algorithm Description**

- The current( $J$ ) produced by the particles is calculated using the equation of

$$J = 2\pi \times F \times \Delta E \times P$$

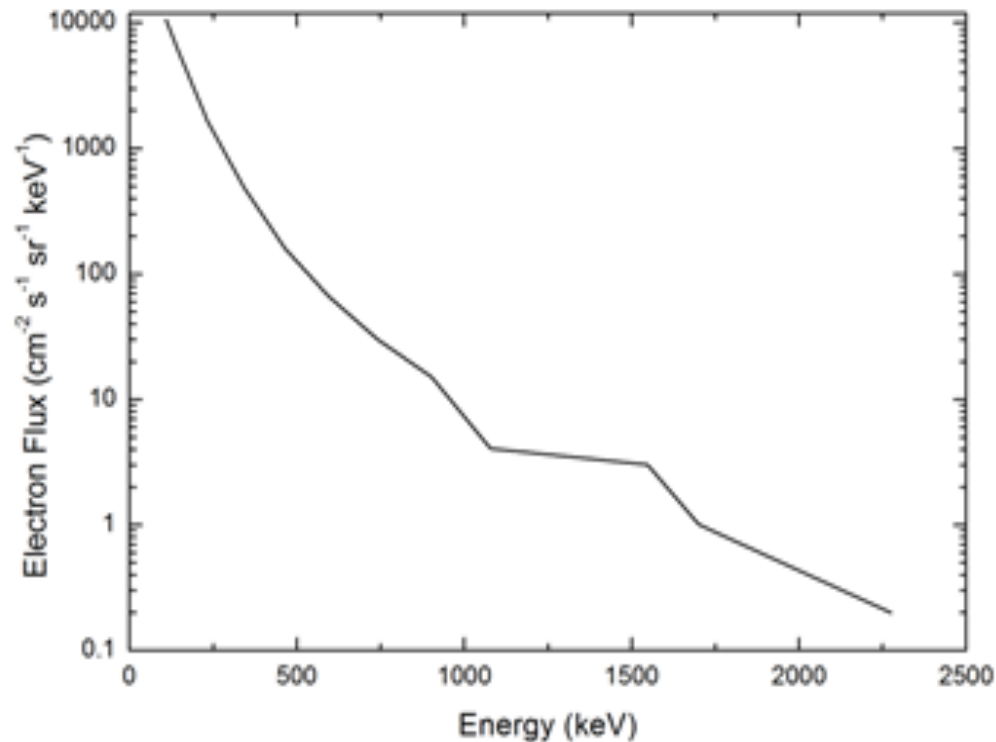
where  $F$  : particle's differential flux,  $E$  : Energy,  
 $P$ : the percentage of particles penetrating the wall

## [Flow Chart of Algorithm]

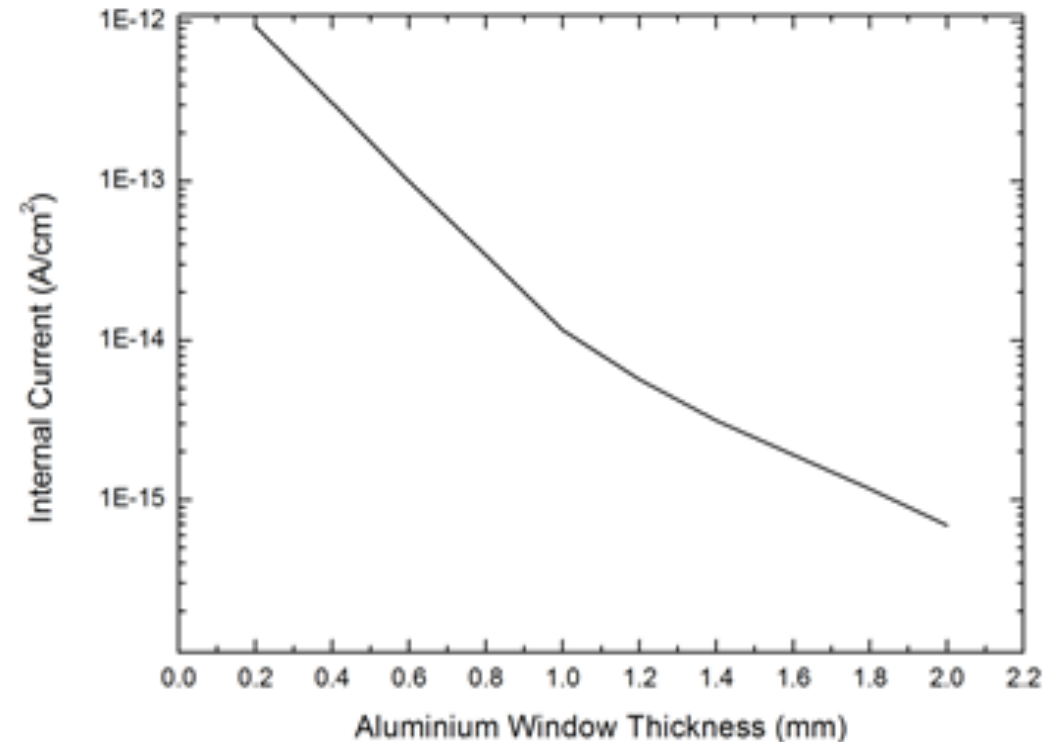


## • Test Result

- Proxy data set is developed using VAP particle data measured at apogee (~30,000km comparable to the geostationary orbit) for algorithm test (Left panel below).
- Internal current with respect to the aluminum thickness is estimated (Right panel below).



[Electron flux measured by VAP mission at 30,000km altitude]



[Internal current with respect to the aluminum thickness]

## • Product Description

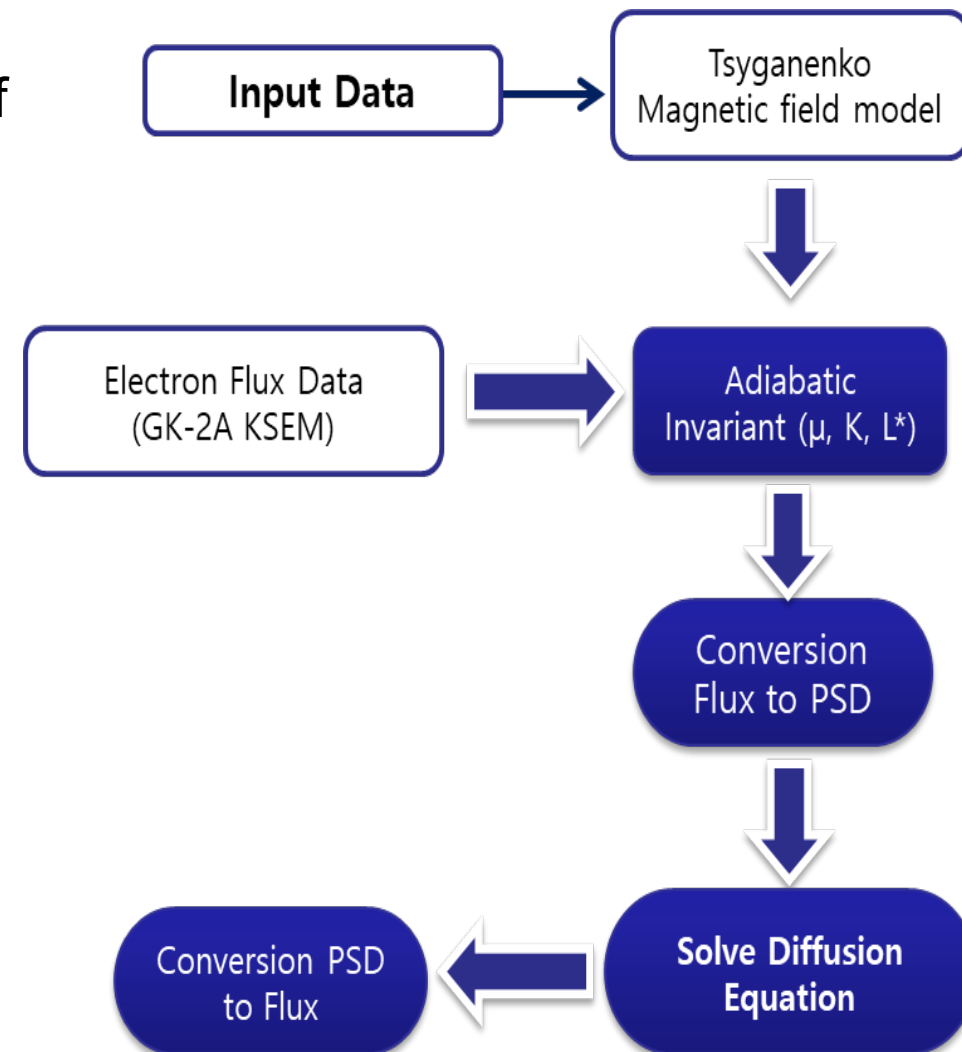
- Real-time electron flux distribution **over whole magnetosphere**(lat, lon, L=2-7) with energy range of a few keV- dozens of MeV.

## • Algorithm Description

- Real-time particle distribution over magnetosphere retrieved by **solving 1-D Fokker-Plank Diffusion**

$$\frac{\partial f}{\partial t} = L^{*2} \frac{\partial}{\partial L^*} \left( \frac{D_{L^*L^*}}{L^{*2}} \frac{\partial f}{\partial L^*} \right) - \frac{f}{\tau_{L^*}} + S(L^*, t)$$

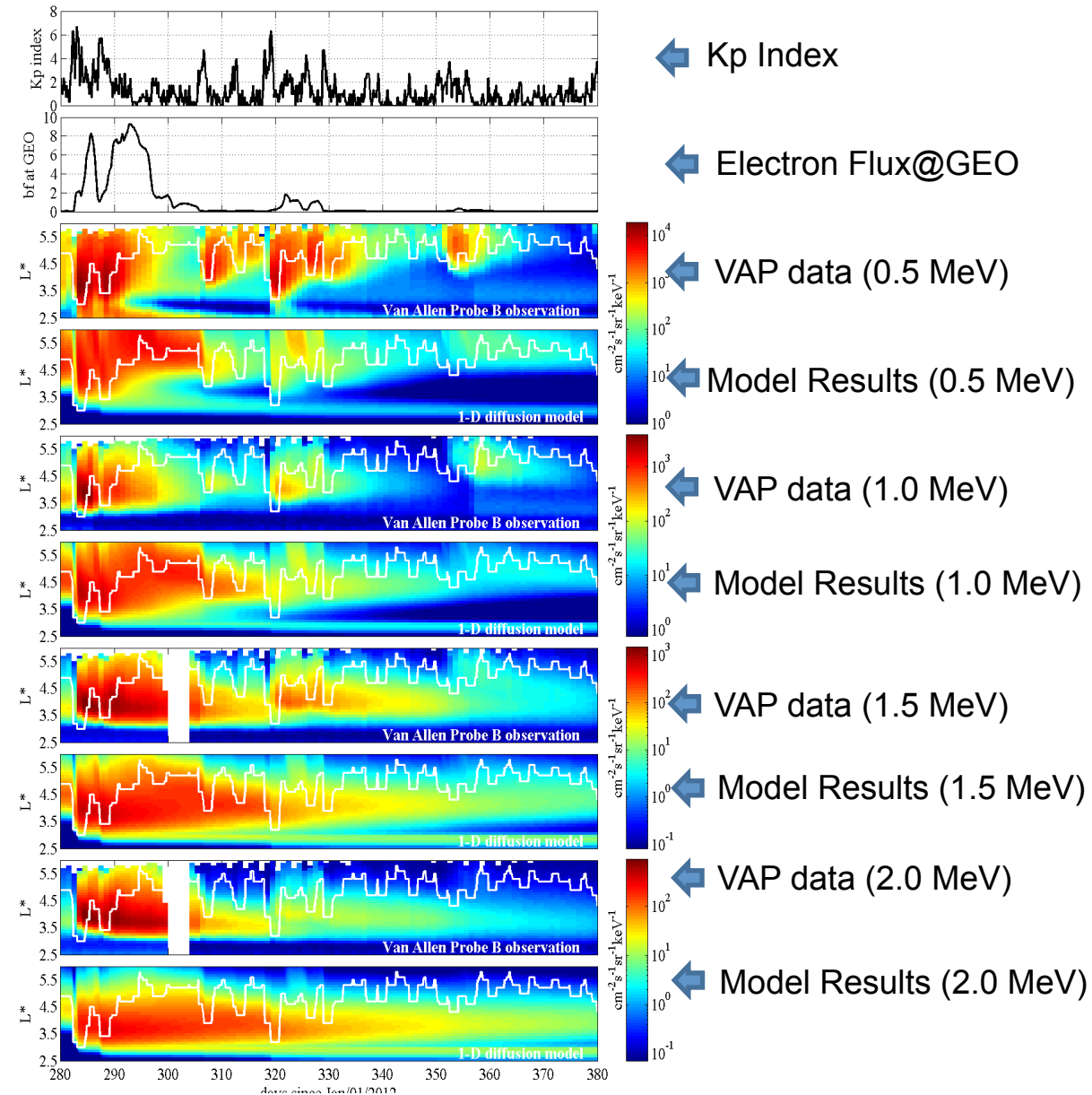
### [Flow Chart of Algorithm]





## • Test Result

- (Upper two panels)  
Time evolution of Kp index and GOES electron fluxes for 100 days since Jan. 01. 2012
- (Lower eight panels)  
Comparison made between fluxes from algorithms with L and energy channels and Van Allen Probes observations(white line)



- **Current Status**

- L2 products developed by KASI (PM : Dr. J. Lee)
- The project is in final phase of algorithm and proto type of codes in test phase

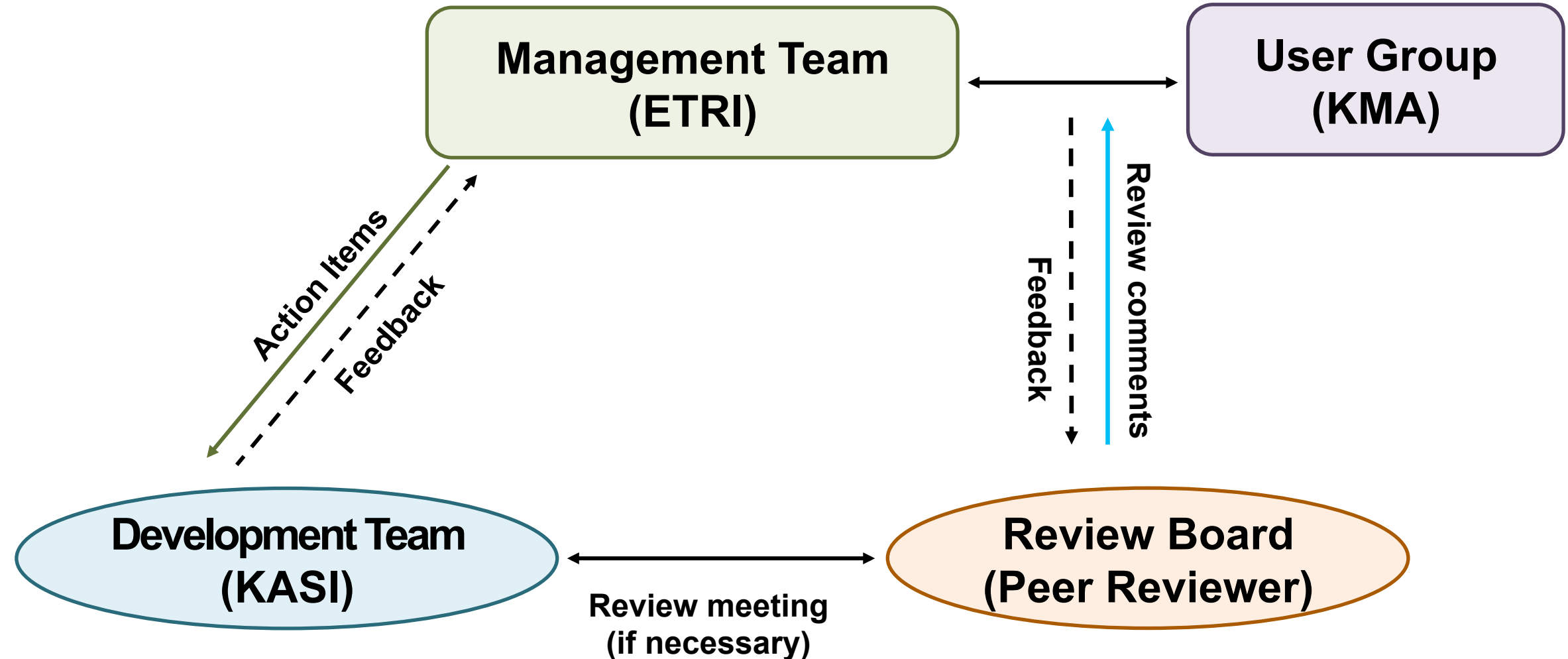
- **Evaluation**

- Collaboration with peer review team for transition of KSEM's L2 products model to operation

- **Goals**

- To secure the performance and operability of five L2 products developed as part of KSEM projects
  - ✓ Real-time Electron flux over whole-magnetosphere from L=2 to 7
  - ✓ 24hr prediction of Electron flux for the targeted satellite orbit
  - ✓ 24hr prediction of Dst
  - ✓ 24hr prediction of Kp
  - ✓ Satellite Charging Index

# III Evaluation Plan

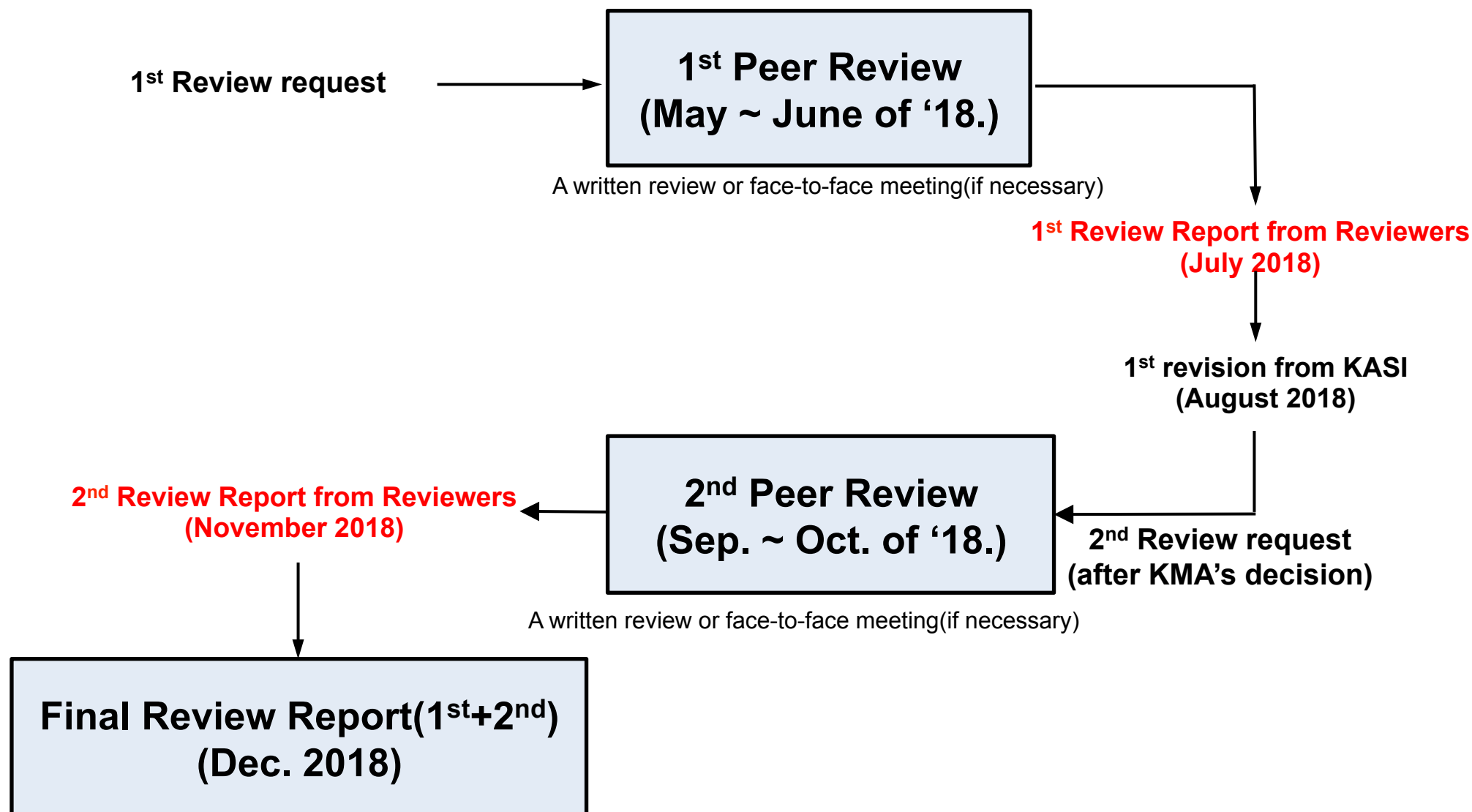


KMA : Korea Meteorological Administration

ETRI : Electronics and Telecommunications Research Institute

KASI : Korea Astronomy and Space Science Institute

# III Evaluation Plan(TBC)





**Thank You**